## Future In-Space Operations Colloquium

## Benefits of a Single-Person Spacecraft for Weigh less Operations



**Brand Griffin** 

August 15, 2012



## Why Look at a Single-Person Spacecraft?

B Griffin

## **Past EVA Contributions**



Samples



Retrieval/ Repair



Future NASA Plans Emphasize Weightless Operation





## **Idea is Not New**

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**Bottle Suit** 

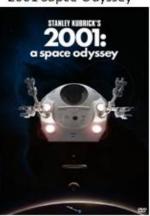
Von Braun





Pod

2001 Sapce Odyssey



1968

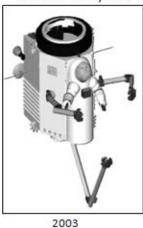
MAWS



1985

SCOUT

Univ. of Maryland



FlexCraft



2010

1954

8/15/2012

MAWSII

Robotnaut arms Howe-Griffin



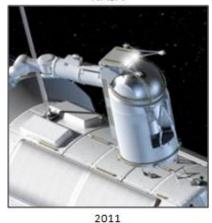
MAWSII Alternative manipulators Howe-Griffin



2010

FlexCraft Cherry Picker

NASA







2010



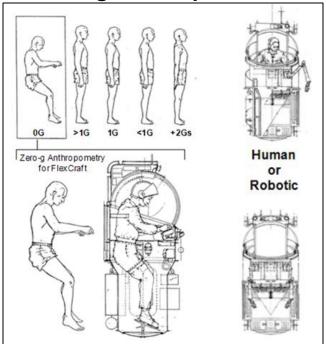
## **Single-Person Spacecraft**

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## **No Reentry**



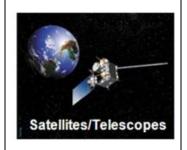
## **Weightless Operations**



## **Venues**



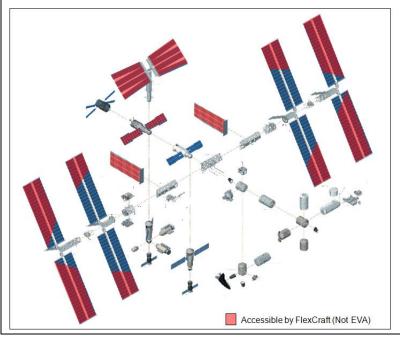




## **ISS Operations**









## **Implications of Low Pressure**

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## User

- Range of motion
- Tactile feedback
- Low forces
- •Minimum fatigue
- Comfort



Design

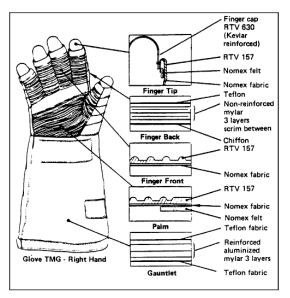
Restraint

Thermal

Puncture

Gas retention

Micrometeoroid



## **ISS Atmosphere** Low pressure space suit 101.3 kPa (14.7 psi) 29.6 kPa (4.3 psi) **Prebreathe** Glove pressure determines suit pressure 29.6 kPa $O_2$ Prebreathe O<sub>2</sub> (4.3psi) 21 % 100% Purge N<sub>2</sub> 101.3 kPa $N_2$ (14.7psi) 79 % Risk of Decompression Sickness "Bends"

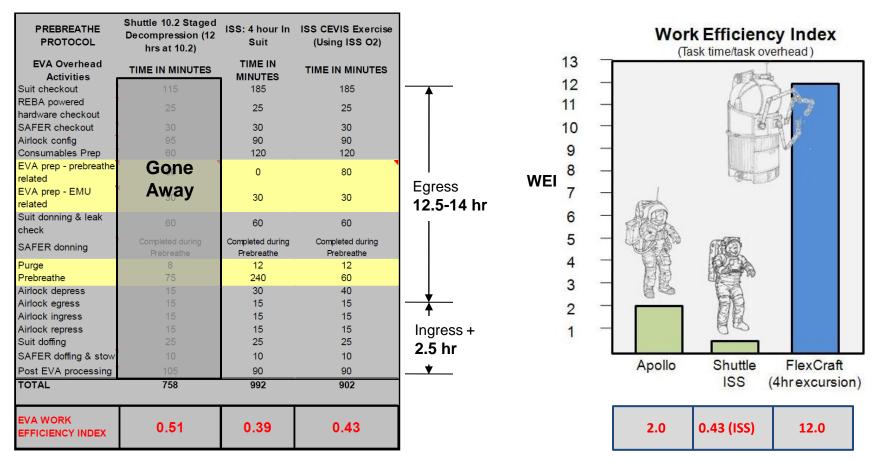


## **Less Overhead Time**

## **Better Work Efficiency Index (WEI)**

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## **EMU EVA Overhead Time**

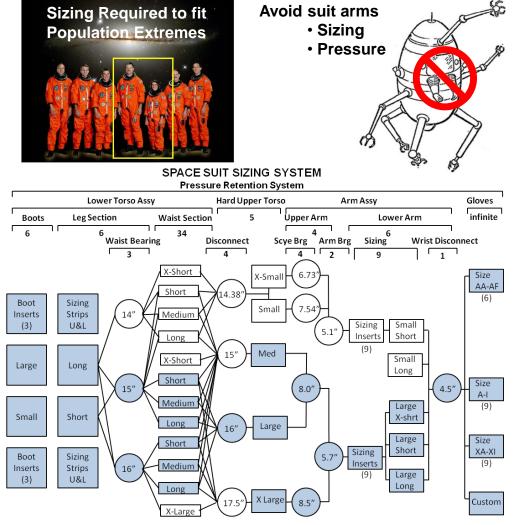


## Direct access without prebreathing minimizes overhead

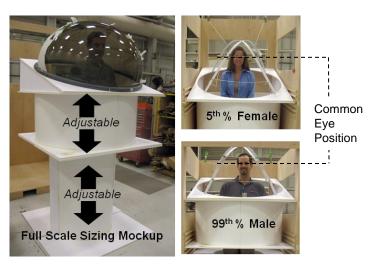


## Many Parts and Adjustments are Required for Proper Suit Fit

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### One Size Fits All





\*Pg 376, US Spacesuits, K. S. Thomas and H. J. McMann



## **Displays and Controls**

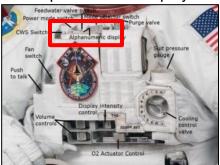
**FlexCraft** 

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## **Space Suit**

Minimal displays (20 character LCD)
Pressure glove dexterity
Out-of-sight controls
Sleeve mounted mirror
Sleeve mounted checklist
Take hand off job to operate
Displays subject to local lighting

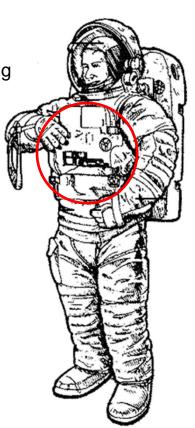
## Alphanumeric Display



**Cuff Check List** 







Multiple displays
(full color flat panels)
Bare hand dexterity
Direct visibility of displays
Operate controls while
manipulators on the job
Cockpit-type environment

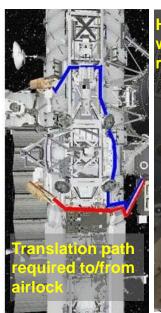




## **Getting to and from the Worksite**

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## Hand-over-hand or SSRMS





## Fly direct to worksite

Emergency return is less than one minute



## Asteroids, Satellite & Telescope

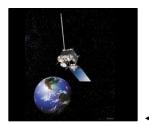


SAFER is for emergency recovery



To be determined:
Propulsion
Tools ?
ORUs
Experiments





Included:
Propulsion
Tools
ORUs
Experiments



## Why Walk When You Can Fly



## **Translation Time**

## **Space Suit**

Translation time is...affected by the crewmembers<sup>1</sup>

- Spacesuit configuration
- Tools carried
- Tethers which must be moved
- The "landscape" over which one is traveling

## **Increment 9 PRCM Replacement EVA**

| Activity                       | Time<br>(min.) |
|--------------------------------|----------------|
| Hatch to Strella               | 9              |
| Translation to PMA1            | 15             |
| Translation to SO              | 5              |
| Tool config, trans to worksite | 14             |
| Stow tools, trans to SO        | 21             |
| Translation to PMA1            | 10             |
| Translation to Piers           | 16             |

Elapsed EVA Translation times

"MMU can return to the airlock from the furthest point on Space Station (about 146.30 m (480 ft)) in less than 1 minute."2

**FlexCraft** 

### MMU ~ FlexCraft Performance

|                  | MMU    | FlexCraft |
|------------------|--------|-----------|
| Delta V (m/s)    | 20*    | 19.5**    |
| Nom. Range (m)   | 137    | same      |
| Operation (hr)   | 6      | same      |
| Propellant       | GN2    | same      |
| Prop mass (kg)   | 5.9    | 14.2      |
| No. Thrusters    | 24     | same      |
| Thrust (N)       | 7.56   | same      |
| Tank Press (kpa) | 20,684 | 23,442    |

<sup>\* 13.7</sup> m/s useful delta v

## Quicker Translation is Safer and Means More Time on the Job

<sup>\*\*</sup> At 450 kg mass

<sup>1</sup> Extravehicular Activity Task Work Efficiency, C Looper and Z. Ney, SAE 2005-01-3014

<sup>2</sup> Role of the Manned Maneuvering Unit for the Space Station, C. E. Whitsett, SAE

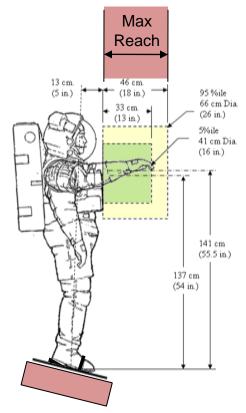


## Foot Restraint Positioning Required for Two Free Hands

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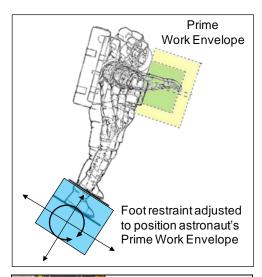
## **Prime Work Envelope**

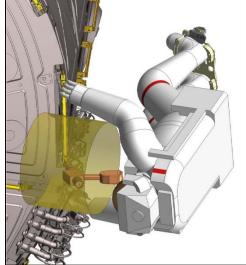
Vision and both hands in same location for all astronauts



**Foot Restraint** 

Allows two hand operations Reaction point for work loads





## ISS

Not all areas EVA accessible
Prescribed translation paths
Requires worksite set up
SSRMS slow (for EVA support 15 cm/sec)
SSRMS not a cherry picker (requires
IVA operations)

## **Asteroid (Near Earth Object)**

Little to no gravity
No translation aids
No EVA propulsion (MMU retired)
Potential suit hazards

## Satellite/Telescope

Only HST EVA serviceable No translation aids Potential suit hazards



## Low Cost Simulator Training with in-Flight Refreshers

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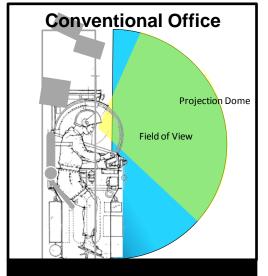
## **Space Suits**



Two Simultaneous Tests.

- Large unique facility
- Many skilled staff
  - **Divers**
  - Suit techs
  - Overhead crane
- Special equipment
- Many test personnel
- Safety issues
- Certification Training
- Pressurized gases
- Control room
- No on-orbit training

## **FlexCraft**





- Conventional Office
- Available projectors
- Available computers
- Simulation software
- •Few test personnel
- No unique training
- No certification
- No safety issues
- On-orbit training
- Laptop platform



## **Grasp Limits and Protection**

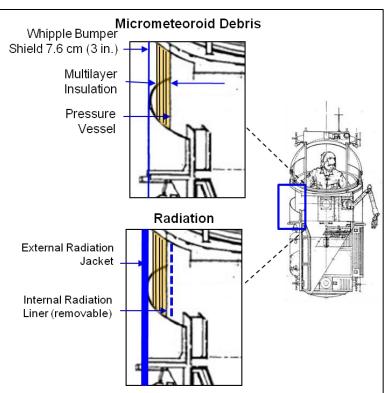
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## **Thermal**

## Even with active heating, touch temperatures are limited to short durations and narrow ranges (-140 to +240°F or -96 to 116°C)<sup>1</sup>

### PHASE VI GLOVE PALM LIMITS, 9 VOLT HEATERS RADIATION, GRASP, AND HIGH PRESSURE GRASP WITH AND WITHOUT HEATING 2 85 LCVG liner (tricot) 80 LCVG water 75 transport tubing 70 TMG cover 65 (ortho-fabric) 1 PSI GRASP 60 HIGH PRESSURE GRASP 55 WINUTES 45 No Known TMG insulation (aluminized **Grasp Limits** 40 35 30 Min. Grasp Limit 30 **EVA Thermal** 25 **Grasp Limits** 20 GH 15 10 Min. Grasp Limit 10 5 120 280 320 3 G. Bue/EC2 analysis suppor TEMPERATURE (°F) -200 deg. F -100 deg. F 160 deg. F 320 deg. F

## Micrometeoroid/Debris and Radiation



- 1. Advanced EVA Roadmaps and Requirements, Richard K. Fullerton, ICES01-2200
- 2. EMU Specification SVHS 7800, Rev BZ, pg 83
- 3. EMU Specification SVHS 7800, Rev BZ, 3.1.1.4.1

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## **Suit-Induced Trauma**

## **Space Suit**

Suit induced trauma can occur even with minimal EVA time<sup>1</sup>



Delaminated Fingernails<sup>2</sup>

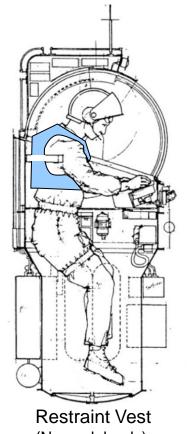


Swelling, Inflammation



Knee abrasion

## **FlexCraft** No trauma anticipated



(No work loads)

<sup>1</sup> Extravehicular Activity - Challenges in Planetary Exploration, Carl Walz / Mike Gernhardt, 27 February, 2008, Third Space Exploration Conference and Exhibit, Denver, CO 2 Opperman RA, Waldie JM, Natapoff A, Newman DJ, Jones JA, Probability of spacesuit-induced fingernail trauma is associated with hand circumference, Aviation Space Environ Med 2010 Oct; 81(10):907-13.

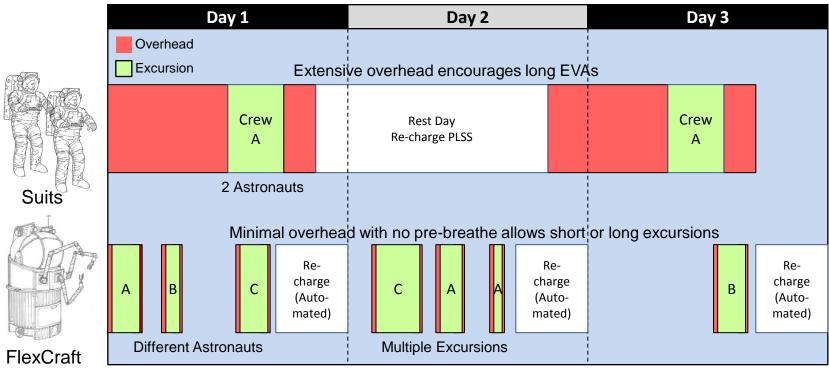


## **Operations Flexibility**

## Mission length with alternating crew members

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## **ISS Operations**



Single-Person Spacecraft:

Short or long excursions
Not physically demanding
No strength or conditioning requirements

No day of rest between EVA No suit re-sizing for different crew No suit cleaning

No prebreathe with automated checkout allows rapid and frequent space access



## Impact of Oxygen Environment

## **Space Suits and Host Spacecraft**

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## Fire and Materials in Different Oxygen Environments

|                                                  | 14.7 psi/21% O <sub>2</sub>                         | 5.2 psi/70-100% O <sub>2</sub>                                                                          |  |
|--------------------------------------------------|-----------------------------------------------------|---------------------------------------------------------------------------------------------------------|--|
| Data Base                                        | 10000 reports                                       | 1000 reports                                                                                            |  |
| Fire Event<br>History                            | High temperature<br>spacecraft events<br>controlled | Apollo 1 (204) &13,<br>hyperbaric chambers                                                              |  |
| Materials<br>Control                             | Minimal                                             | Significant, beta bagging,<br>nonflammable<br>paints/coatings, aluminum<br>foil tape, containment, etc. |  |
| Materials<br>Selection                           | Numerous options                                    | Severely restricted                                                                                     |  |
| Off-the-shelf<br>Hardware                        | Minimal modifications                               | Significant design changes,<br>testing/analysis*                                                        |  |
| Fire Detection                                   | Standard                                            | Rapid events                                                                                            |  |
| Fire<br>Suppression                              | Standard techniques effective                       | Ineffective                                                                                             |  |
| Propagation<br>Potential/Self-<br>Extinguishment | Most materials self-<br>extinguish                  | Few materials self-<br>extinguish, rapid propagation                                                    |  |

Dr. Michael D. Pedley, NASA/JSC/ES5, Dennis E. Griffin, NASA/MSFC/EH02

# 1980 Oxygen Suit Fire



**Materials Availability** NHB 8060, 1 Flammability Test Data Cumulation Percent of Polymeric Materials Passing 80 70 -Values are very approximate 60 Dr. Michael Pedley JSC/SMD Sept. 1992 20 -10 -70 30 20.9 Percentage Oxygen

Low Pressure Spiral
Glove Drives Host Requirements

1
2
3

<sup>\*</sup> Examples: (from Designing for EVA Applications, Gene Lutz, EVA Systems Project Office. Johnson Space Center, 13 February 2007)

<sup>-</sup> A \$150 Polar heart rate monitor may require \$200K (or more) of engineering testing, analysis and certification before it can be used in the Space Suit

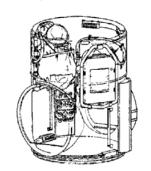
<sup>-</sup> An emergency room EKG may require significant modifications to ensure the electrode leads do not detach from the crewmember in the Space Suit



## **Mass Comparison**

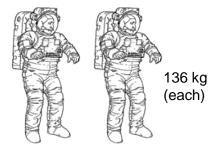
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**1012 kg** (with airlock) **844 kg** (without airlock (Orion), 4 EMUs)

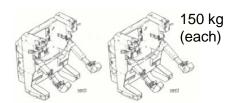


440 kg

Shuttle Airlock



Extravehicular Mobility Unit (EMU)



Manned Maneuvering Unit (MMU) (Representative, no longer available)

**452 kg** (Wet)



| Category              | Mass (kg) |
|-----------------------|-----------|
| Structures            | 121       |
| Propulsion            | 51        |
| Power                 | 42        |
| Avionics              | 40        |
| Thermal               | 21        |
| ECLSS                 | 44        |
| Docking Mechanism     | 20        |
| GROWTH                | 41        |
| DRY MASS              | 379       |
| Non-Prop Fluids       | 1         |
| Manipulators          | 58        |
| INERT MASS            | 437       |
| Total Less Propellant | 437       |
| Propellant            | 14        |
| TOTAL GROSS MASS      | 452       |

**460 kg** (Dry)



The 1965 Cessna 150E 1,010 lb (460 kg)

**286 kg** (Dry with std equip)



New Skykit Savannah 630 lb (286 kg)



## **Step Function Capability**

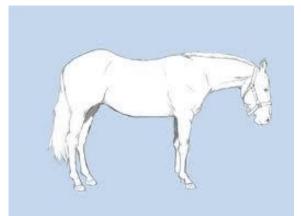
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"If I asked the people what they wanted, they would have said faster horses."



Henry Ford

## Faster Horses?





Better "Weightless" Suits?

## New Capability





New Capability